Paper / Subject Code: 50703 / Fluid Flow (FF)

20-Nov-2019 1T00523 - S.E.(Chemical Engineering)(SEM-III)(Choice Base) / 50703 - Fluid Flow (FF) 77301

Time: 3 Hours Marks: 80

N. B.: (1) Question No. 1 is compulsory.

- (2) Attempt any three questions from remaining five questions.
- (3) Assume suitable data if necessary.

Q. 1 Answer any Four questions

[20]

- a) Derive Newton's law of viscosity and mention the various units of viscosity.
- b) A plate of 0.05 mm at a distance from a fixed plate moves at 1.2 m/s and requires a force of 2.2 N/m² to maintain this speed. Find the viscosity of the fluid between the plates.
- c) Derive equation of continuity for compressible and incompressible fluids.
- d) What do you mean by compressible fluids and explain what are the objectives of learning compressible flow in short.
- e) Write the classification of pressure measurement devices.

0.2

(a) Explain the different types of fluid flow. (any five types)

S [10]

[10]

- (b) Oil of viscosity 0.098 kg/(m.s.)and sp.gr 0.9 flows through a horizontal pipe of 2.5 cm diameter. If the pressure drop per meter length of pipe is 0.12 kgf/cm². Determine (i) The rate of flow.(ii) Reynolds number.
 - (iii) The power required per 50 m length pipe to maintain flow.

Q. 3

- (a) Water is flowing through a pipe having diameters 30 cm and 50 cm at the bottom and upper end respectively. The intensity of pressure at the bottom end is 3 kgf/cm² and at the upper end is 1.5 kgf/cm². Determine the difference in datum head if the rate of flow through pipe is 50 lit/sec. [10]
- (b) Draw and explain the propagation of pressure waves, when Ma = 1, Ma<1 and Ma>1

Q. 4

- (a) Derive an expression for the velocity distribution, shear stress distribution and relation between average velocity and maximum velocity for the laminar flow of fluid through the circular pipe. [10]
- (b) Explain any two types of valves with neat sketch. [10]

Q. 5

- (a) Find the Mach number when an aeroplane is flying at 1100 km/hr through still air having a pressure of 7 N/cm² and temperature -5°C, wind velocity may be taken as zero.

 Take R = 287.14 J/kg. °K. Calculate the pressure temperature and density of air at stagnation point on the nose of the plane. Take k = 1.4
- (b) State the different types of pumps and explain the centrifugal pump with neat sketch. [10]

Q. 6

- (a) Derive an expression for hydrostatic equilibrium [05]
 (b) Define and explain drag force. [05]
- (c) Define and give the significance of Mach Number. [05]

(d) Explain NPSHR and NPSHA [05]

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